

# Vibration 3

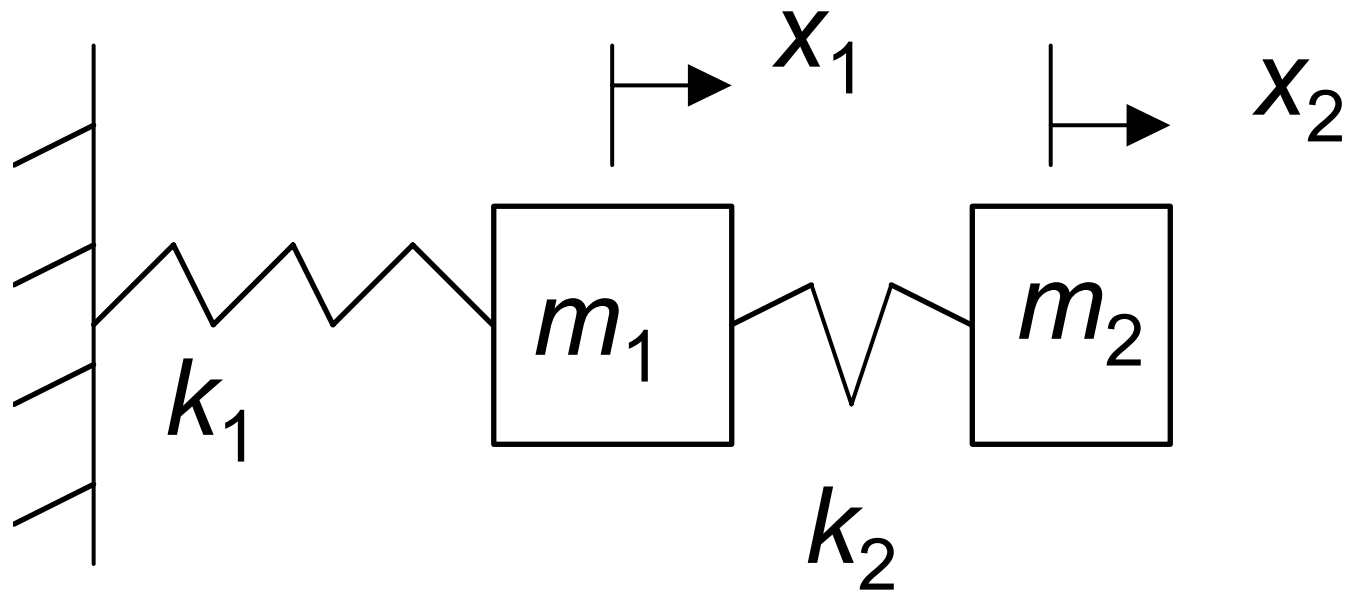
## **2DOF and Multiple DOF systems**

# Chapter 4 Multiple Degree of Freedom Systems

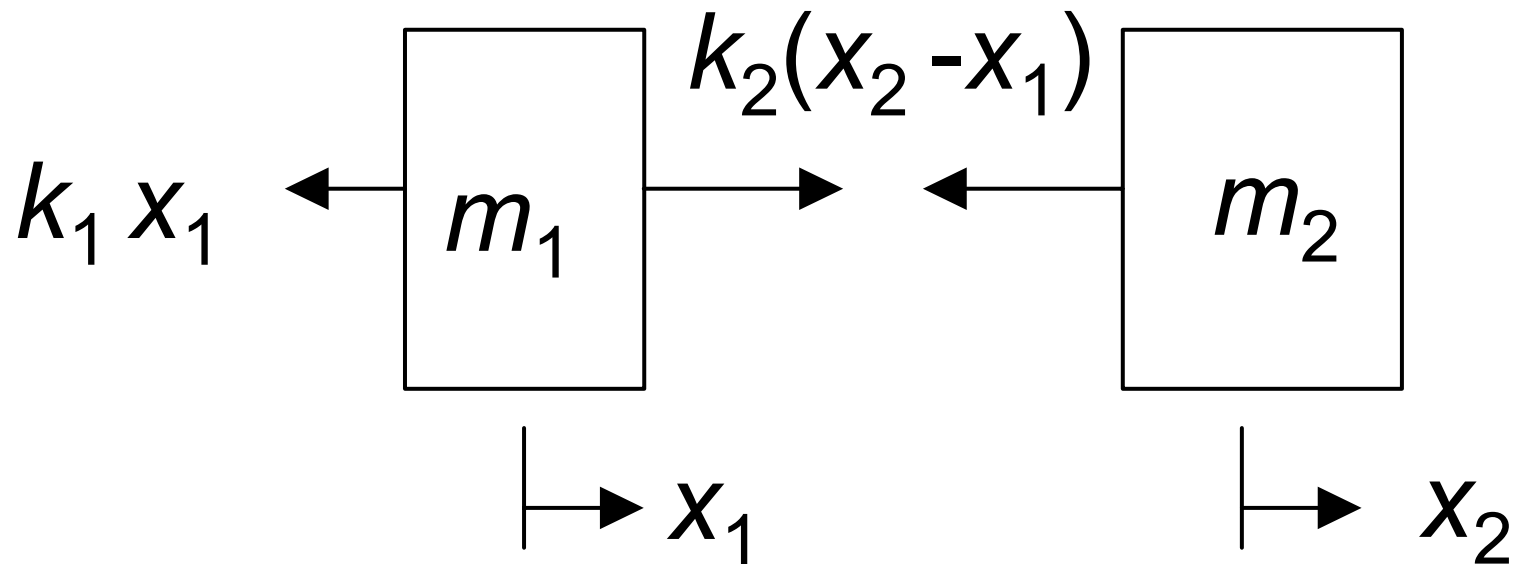
Extending the first 3 chapters to more than one degree of freedom

Rao's textbook: Chap 5

# Two Degrees of Freedom (4.1)



# Free-Body Diagram



# Equations of Motion

$$m_1 \ddot{x}_1(t) = -k_1 x_1(t) + k_2 (x_2(t) - x_1(t))$$

$$m_2 \ddot{x}_2(t) = -k_2 (x_2(t) - x_1(t))$$

Rearranging :

$$m_1 \ddot{x}_1(t) + (k_1 + k_2) x_1(t) - k_2 x_2(t) = 0$$

$$m_2 \ddot{x}_2(t) - k_2 x_1(t) + k_2 x_2(t) = 0$$

# Initial Conditions

- Two **coupled**, **second -order**, ordinary differential equations with constant coefficients
- Needs 4 constants of integration to solve
- Thus **4 initial conditions** on positions and velocities

$$x_1(0) = x_{10}, \dot{x}_1(0) = \dot{x}_{10}, x_2(0) = x_{20}, \dot{x}_2(0) = \dot{x}_{20}$$

# Solution by Matrix Methods

$$\mathbf{x}(t) = \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix}, \dot{\mathbf{x}}(t) = \begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \end{bmatrix}, \ddot{\mathbf{x}}(t) = \begin{bmatrix} \ddot{x}_1(t) \\ \ddot{x}_2(t) \end{bmatrix}$$

$$\mathbf{M} = \begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix}, \mathbf{K} = \begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix}$$

$$\mathbf{M}\ddot{\mathbf{x}} + \mathbf{K}\mathbf{x} = \mathbf{0}$$

# Initial Conditions

$$\mathbf{x}(0) = \begin{bmatrix} x_{10} \\ x_{20} \end{bmatrix}, \quad \dot{\mathbf{x}}(0) = \begin{bmatrix} \dot{x}_{10} \\ \dot{x}_{20} \end{bmatrix}$$



## Solution:

$$\text{Let } \mathbf{x}(t) = \mathbf{u}e^{j\omega t}$$

$$j = \sqrt{-1}, \quad \mathbf{u} \neq \mathbf{0}, \quad \omega \text{ unknown}$$

$$\Rightarrow (-\omega^2 M + K)\mathbf{u}e^{j\omega t} = \mathbf{0}$$

$$\Rightarrow (-\omega^2 M + K)\mathbf{u} = \mathbf{0}$$

## Changes ODE into algebraic equation

$$(-\omega^2 M + K)\mathbf{u} = \mathbf{0} \Rightarrow$$

two algebraic equation in 3 unknowns

$$\mathbf{u} = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}, \text{ and } \omega$$

# Condition for Solution:

$\text{inv}(-\omega^2 M + K)$  exists  $\Rightarrow \mathbf{u} = \mathbf{0}$

Require  $\mathbf{u} \neq \mathbf{0} \Rightarrow (-\omega^2 M + K)^{-1}$  does not exist

or  $\det(-\omega^2 M + K) = 0$

One equation in one unknown  $w$

## Back to our specific system: the characteristic equation

$$\det(-\omega^2 M + K) = 0 \Rightarrow$$

$$\det \begin{bmatrix} -\omega^2 m_1 + k_1 + k_2 & -k_2 \\ -k_2 & -\omega^2 m_2 + k_2 \end{bmatrix} = 0 \Rightarrow$$

$$m_1 m_2 \omega^4 - (m_1 k_2 + m_2 k_1 + m_2 k_2) \omega^2 + k_1 k_2 = 0$$

**Quadratic in  $\omega^2$  so four solutions**  
 **$\omega^2_1$  and  $\omega^2_2$  or  $\pm\omega_1$  and  $\pm\omega_2$**

## Calculating the corresponding vectors $u_1$ and $u_2$

A vector equation for each square frequency

$$(-\omega_1^2 M + K)\mathbf{u}_1 = \mathbf{0}$$

And:

$$(-\omega_2^2 M + K)\mathbf{u}_2 = \mathbf{0}$$

4 equations in the 4 unknowns (each vector has 2 components, but...)

## Numerical examples

- $m_1=9$  kg,  $m_2=1$ kg,  $k_1=24$  N/m and  $k_2=3$  N/m
- Characteristic equation becomes

$$\omega^4 - 6\omega^2 + 8 = (\omega^2 - 2)(\omega^2 - 4) = 0$$

$$\omega^2 = 2 \text{ and } \omega^2 = 4 \text{ or}$$

$$\omega_{1,3} = \pm\sqrt{2} \text{ rad/s} \quad \omega_{2,4} = \pm 2 \text{ rad/s}$$

Each value of  $\omega^2$  yields an expression for  $\mathbf{u}$ :

# Computing the vectors $\mathbf{u}$

For  $\omega_1^2 = 2$ , let  $\mathbf{u}_1 = \begin{bmatrix} u_{11} \\ u_{12} \end{bmatrix}$  then we have

$$(-\omega_1^2 M + K)\mathbf{u} = \mathbf{0} \Rightarrow$$

$$\begin{bmatrix} 27 - 9(2) & -3 \\ -3 & 3 - (2) \end{bmatrix} \begin{bmatrix} u_{11} \\ u_{12} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \Rightarrow$$

$$9u_{11} - 3u_{12} = 0 \quad \text{and} \quad -3u_{11} + u_{12} = 0$$

**2 equations, 2 unknowns but DEPENDENT!**

## continued

$$\frac{u_{11}}{u_{12}} = \frac{1}{3} \Rightarrow u_{11} = \frac{1}{3}u_{12} \text{ from both equations :}$$

only the direction, not the magnitude can be determined!

This is because :  $\det(-\omega_1^2 M + K) = 0$ .

The magnitude is arbitrary. Suppose  $\mathbf{u}_1$  satisfies

$(-\omega_1^2 M + K)\mathbf{u}_1 = \mathbf{0}$ , so does  $a\mathbf{u}_1$ ,  $a$  arbitrary :

$$(-\omega_1^2 M + K)a\mathbf{u}_1 = \mathbf{0} \Leftrightarrow (-\omega_1^2 M + K)\mathbf{u}_1 = \mathbf{0}$$



Likewise for the second value of  $\omega^2$ :

For  $\omega_2^2 = 4$ , let  $\mathbf{u}_2 = \begin{bmatrix} u_{21} \\ u_{22} \end{bmatrix}$  then we have

$$(-\omega_1^2 M + K)\mathbf{u} = \mathbf{0} \Rightarrow$$

$$\begin{bmatrix} 27 - 9(4) & -3 \\ -3 & 3 - (4) \end{bmatrix} \begin{bmatrix} u_{21} \\ u_{22} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \Rightarrow$$

$$-9u_{21} - 3u_{22} = 0 \quad \text{or} \quad u_{21} = -\frac{1}{3}u_{22}$$

Note that the other equation is the same

# What to do about the magnitude!

Several possibilities, here we just fix one element:

Choose:

$$u_{12} = 1 \Rightarrow \mathbf{u}_1 = \begin{bmatrix} 1/3 \\ 1 \end{bmatrix}$$

Choose:

$$u_{22} = 1 \Rightarrow \mathbf{u}_2 = \begin{bmatrix} -1/3 \\ 1 \end{bmatrix}$$

Thus the solution to the algebraic matrix equation is:

$$\omega_{1,3} = \pm\sqrt{2}, \quad \mathbf{u}_1 = \begin{bmatrix} 1/3 \\ 1 \end{bmatrix}$$

$$\omega_{2,4} = \pm 2, \quad \mathbf{u}_2 = \begin{bmatrix} -1/3 \\ 1 \end{bmatrix}$$

## Return now to the time response:

We have four solutions:

$$\mathbf{x}(t) = \mathbf{u}_1 e^{-j\omega_1 t}, \mathbf{u}_1 e^{j\omega_1 t}, \mathbf{u}_2 e^{-j\omega_2 t}, \mathbf{u}_2 e^{j\omega_2 t} \Rightarrow$$

Since linear we can combine as:

$$\begin{aligned}\mathbf{x}(t) &= a\mathbf{u}_1 e^{-j\omega_1 t} + b\mathbf{u}_1 e^{j\omega_1 t} + c\mathbf{u}_2 e^{-j\omega_2 t} + d\mathbf{u}_2 e^{j\omega_2 t} \\ \Rightarrow \mathbf{x}(t) &= (ae^{-j\omega_1 t} + be^{j\omega_1 t})\mathbf{u}_1 + (ce^{-j\omega_2 t} + de^{j\omega_2 t})\mathbf{u}_2 \\ &= \underline{A_1 \sin(\omega_1 t + \phi_1)\mathbf{u}_1 + A_2 \sin(\omega_2 t + \phi_2)\mathbf{u}_2}\end{aligned}$$

where  $A_1, A_2, \phi_1$ , and  $\phi_2$  are constants of integration  
determined by initial conditions

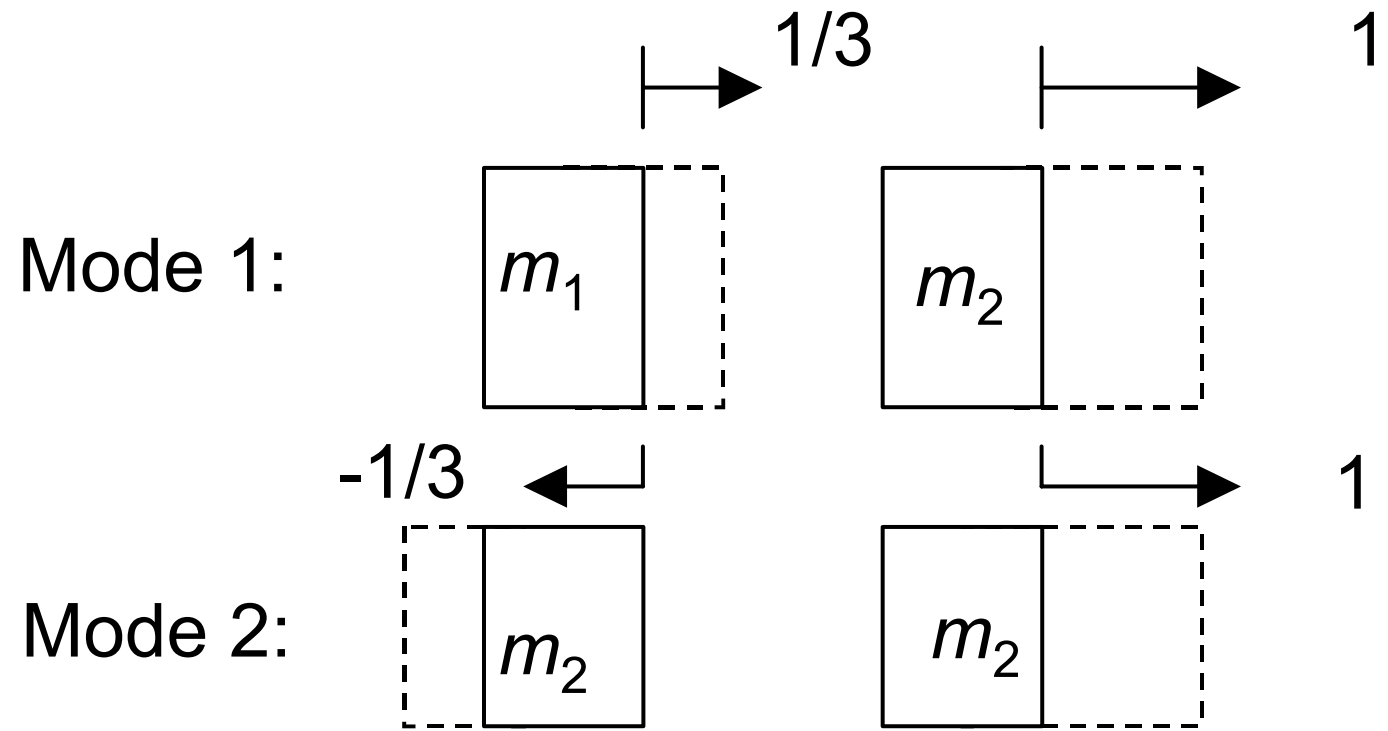
# Physical interpretation of all that math!

- Each of the **TWO** masses is oscillating at **TWO** natural frequencies  $\omega_1$  and  $\omega_2$
- The relative magnitude of each sine term, and hence of the magnitude of oscillation of  $m_1$  and  $m_2$  is determined by the value of  $A_1\mathbf{u}_1$  and  $A_2\mathbf{u}_2$
- The vectors  $\mathbf{u}_1$  and  $\mathbf{u}_2$  are called **mode shapes**

# What is a mode shape?

- First note that  $A_1, A_2, \phi_1$  and  $\phi_2$  are determined by the initial conditions
- Choose them so that  $A_2 = \phi_1 = \phi_2 = 0$
- Then:
$$\mathbf{x}(t) = \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} = A_1 \begin{bmatrix} u_{11} \\ u_{12} \end{bmatrix} \sin \omega_1 t$$
- Thus each mass oscillates at (one) frequency  $\omega_1$  with magnitudes proportional to  $\mathbf{u}_1$  the 1st *mode shape*

# Mode shapes:



## Example (continue on)

consider  $\mathbf{x}(0) = \begin{bmatrix} 1 \\ 0 \end{bmatrix}$  mm,  $\dot{\mathbf{x}}(0) = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$

$$\begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} = \begin{bmatrix} \frac{A_1}{3} \sin(\sqrt{2}t + \phi_1) - \frac{A_2}{3} \sin(2t + \phi_2) \\ A_1 \sin(\sqrt{2}t + \phi_1) + A_2 \sin(2t + \phi_2) \end{bmatrix}$$

$$\begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \end{bmatrix} = \begin{bmatrix} \frac{A_1}{3} \sqrt{2} \cos(\sqrt{2}t + \phi_1) - \frac{A_2}{3} 2 \cos(2t + \phi_2) \\ A_1 \sqrt{2} \cos(\sqrt{2}t + \phi_1) + A_2 2 \cos(2t + \phi_2) \end{bmatrix}$$



At  $t=0$  we have

$$\begin{bmatrix} 1 \text{ mm} \\ 0 \end{bmatrix} = \begin{bmatrix} \frac{A_1}{3} \sin(\phi_1) - \frac{A_2}{3} \sin(\phi_2) \\ A_1 \sin(\phi_1) + A_2 \sin(\phi_2) \end{bmatrix}$$

$$\begin{bmatrix} 0 \\ 0 \end{bmatrix} = \begin{bmatrix} \frac{A_1}{3} \sqrt{2} \sin(\phi_1) - 2 \frac{A_2}{3} \sin(\phi_2) \\ A_1 \sqrt{2} \sin(\phi_1) + 2A_2 \sin(\phi_2) \end{bmatrix}$$

4 equations in 4 unknowns:

$$3 = A_1 \sin(\phi_1) - A_2 \sin(\phi_2)$$

$$0 = A_1 \sin(\phi_1) + A_2 \sin(\phi_2)$$

$$0 = A_1 \sqrt{2} \cos(\phi_1) - A_2 2 \cos(\phi_2)$$

$$0 = A_1 \sqrt{2} \cos(\phi_1) + A_2 2 \cos(\phi_2)$$

Yields:

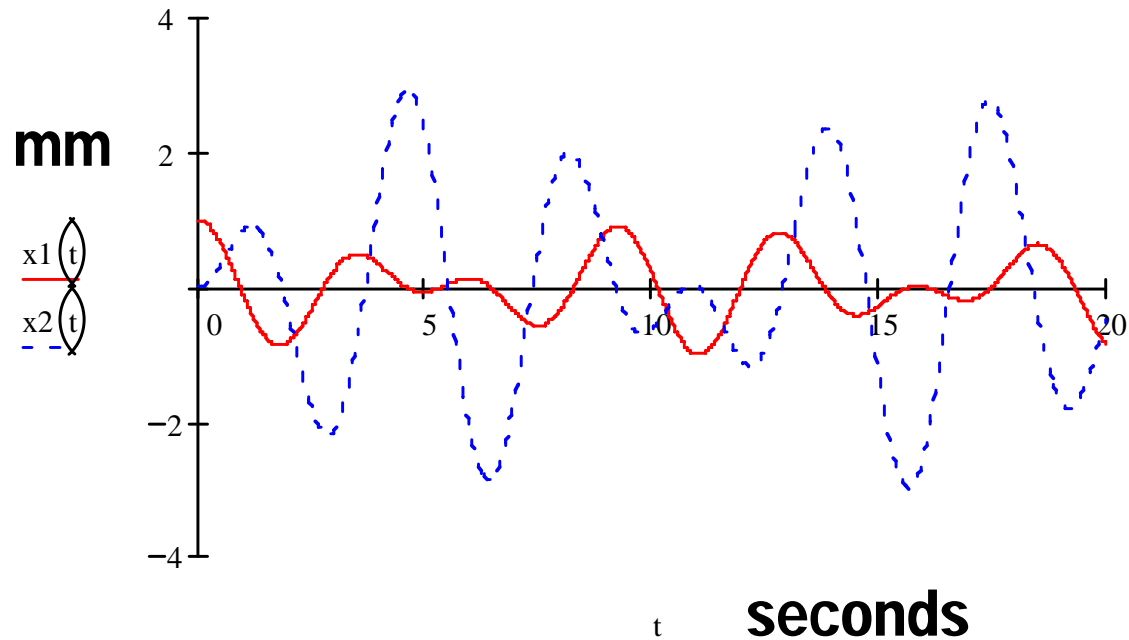
$$A_1 = 1.5 \text{ mm}, A_2 = -1.5 \text{ mm}, \phi_1 = \phi_2 = \frac{\pi}{2} \text{ rad}$$

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## Solution:

$$x_1(t) = 0.5 \cos \sqrt{2}t + 0.5 \cos 2t$$

$$x_2(t) = 1.5 \cos \sqrt{2}t - 1.5 \cos 2t$$



# Solution as a sum of modes

$$\mathbf{x}(t) = \mathbf{u}_1 \cos \omega_1 t + \mathbf{u}_2 \cos \omega_2 t$$

Determines how the first frequency contributes to the response

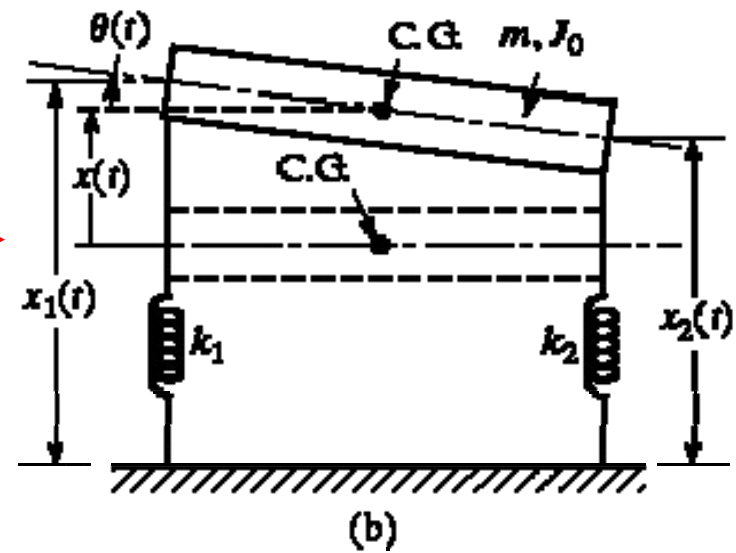
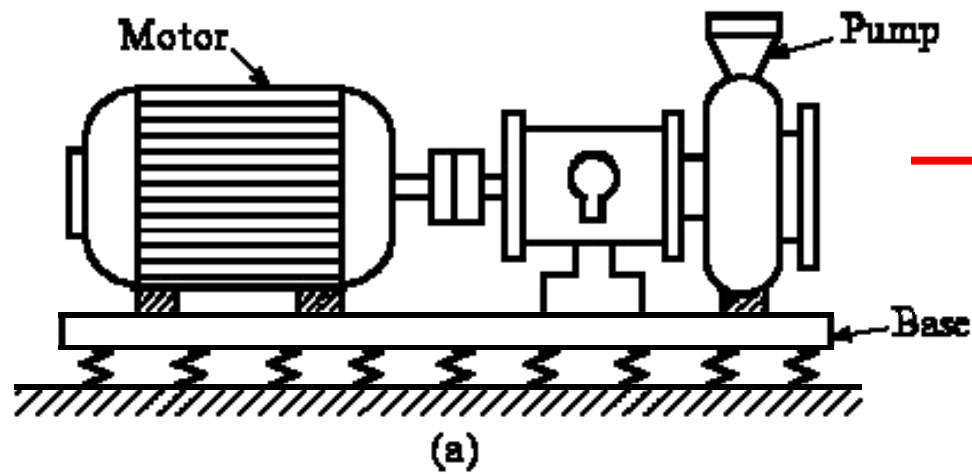
Determines how the second frequency contributes to the response

# Things to note

- Two degrees of freedom implies **two** natural frequencies
- Each mass oscillates at with these two frequencies present in the response
- Frequencies are not those of two component systems

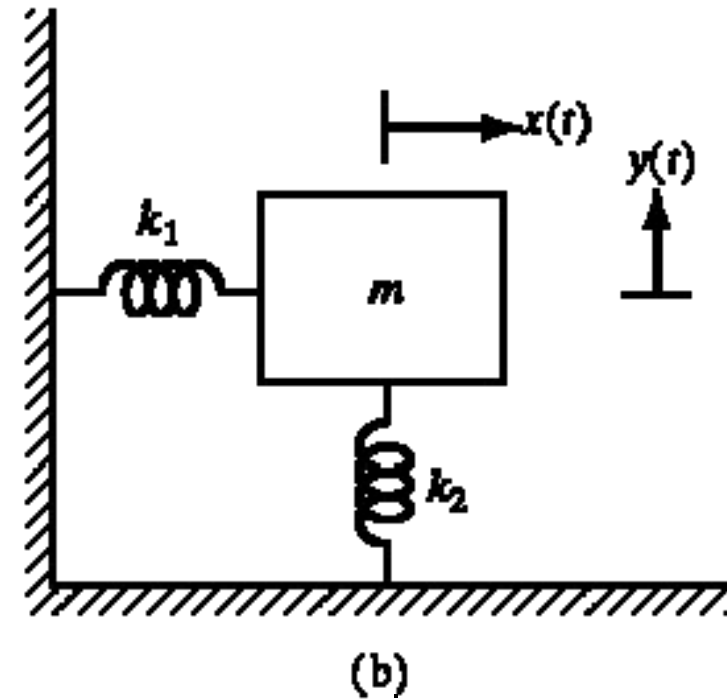
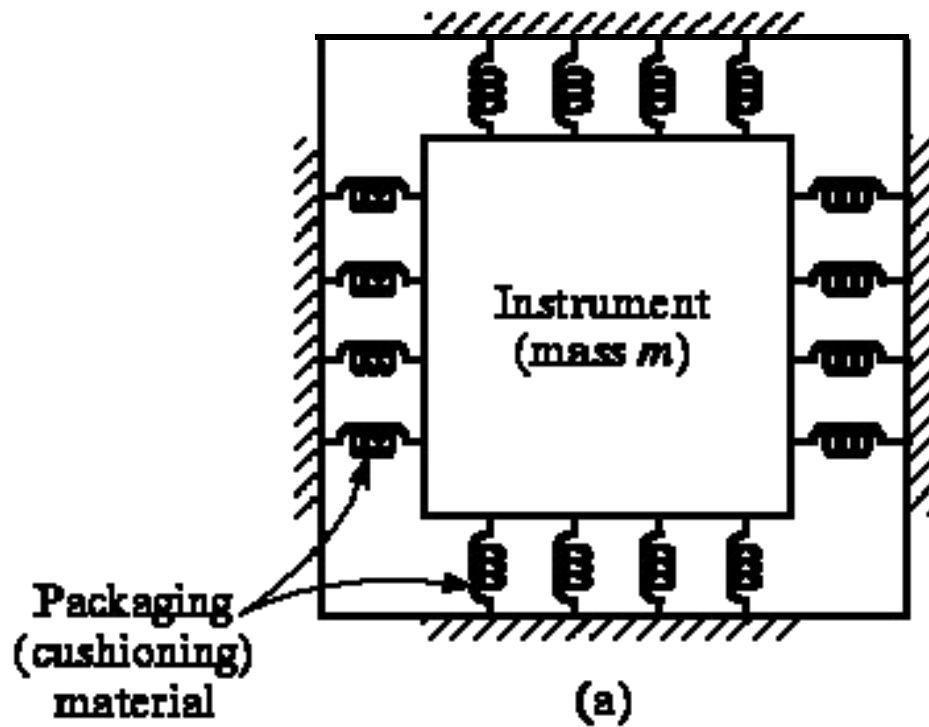
$$\omega_1 = \sqrt{2} \neq \sqrt{\frac{k_1}{m_1}} = 1.63, \omega_2 = 2 \neq \sqrt{\frac{k_2}{m_2}} = 1.732$$

# Motor-pump system on springs



Figs.5.1

## Packaging of an instrument (portable electronics)



Figs.5.2

## Principle coordinates

- As is evident from the systems shown in Figs.5.1 and 5.2, the configuration of a system can be specified by a set of independent coordinates termed as **generalized coordinates**, such as length, angle, or some other physical parameters.
- *Principle coordinates* is defined as any set of coordinates that leads a coupled equation of motion to an **uncoupled** system of equations.



## Equations of Motion for 2DOF System

$$m_1 \ddot{x}_1 + (c_1 + c_2) \dot{x}_1 - c_2 \dot{x}_2 + (k_1 + k_2)x_1 - k_2 x_2 = F_1 \quad (5.1)$$

$$m_2 \ddot{x}_2 - c_2 \dot{x}_1 + (c_2 + c_3) \dot{x}_2 - k_2 x_1 + (k_2 + k_3)x_2 = F_2 \quad (5.2)$$

$$[m] \ddot{\vec{x}}(t) + [c] \dot{\vec{x}}(t) + [k] \vec{x}(t) = \vec{F}(t) \quad (5.3)$$

where  $[m]$ ,  $[c]$ , and  $[k]$  are called the mass, damping, and stiffness matrices, respectively, and are given by

## Properties of M & K Matrices

$$[m] = \begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \quad [k] = \begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_2 + k_3 \end{bmatrix}$$

$$[m]^T = [m], \quad [k]^T = [k]$$

where the superscript T denotes the transpose of the matrix.

Matrices [m] and [k] are **symmetric**:

## Ex 5.3 Free Vibration Response of a Two Degree of Freedom System

Find the free vibration response of the system shown in Fig.5.3(a) with  $k_1 = 30$ ,  $k_2 = 5$ ,  $k_3 = 0$ ,  $m_1 = 10$ ,  $m_2 = 1$  and  $c_1 = c_2 = c_3 = 0$  for the initial conditions  $x_1(0) = 1$ ,  $\dot{x}_1(0) = x_2(0) = \dot{x}_2(0)$ .

*Solution:* For the given data, the eigenvalue problem, Eq.(5.8), becomes

$$\begin{bmatrix} -m_1\omega^2 + k_1 + k_2 & -k_2 \\ -k_2 & -m_2\omega^2 + k_2 + k_3 \end{bmatrix} \begin{Bmatrix} X_1 \\ X_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

**or**

$$\begin{bmatrix} -10\omega^2 + 35 & -5 \\ -5 & -\omega^2 + 5 \end{bmatrix} \begin{Bmatrix} X_1 \\ X_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix} \quad (\text{E.1})$$

## Example 5.3 Solution

By setting the determinant of the coefficient matrix in Eq.(E.1) to zero, we obtain the frequency equation,

$$10\omega^4 - 85\omega^2 + 150 = 0 \quad (\text{E.2})$$

from which the natural frequencies can be found as

$$\begin{aligned} \omega_1^2 &= 2.5, & \omega_2^2 &= 6.0 \\ \omega_1 &= 1.5811, & \omega_2 &= 2.4495 \end{aligned} \quad (\text{E.3})$$

The **normal modes** (or **eigenvectors**) are given by

$$X^{(1)} \begin{Bmatrix} X_1^{(1)} \\ X_2^{(1)} \end{Bmatrix} = \begin{Bmatrix} 1 \\ 2 \end{Bmatrix} X_1^{(1)} \quad (\text{E.4})$$

$$X^{(2)} \begin{Bmatrix} X_1^{(2)} \\ X_2^{(2)} \end{Bmatrix} = \begin{Bmatrix} 1 \\ -5 \end{Bmatrix} X_1^{(2)} \quad (\text{E.5})$$

## Example 5.3 Solution

The free vibration responses of the masses  $m_1$  and  $m_2$  are given by (see Eq.5.15):

$$x_1(t) = X_1^{(1)} \cos(1.5811t + \phi_1) + X_1^{(2)} \cos(2.4495t + \phi_2) \quad (\text{E.6})$$

$$x_2(t) = 2X_1^{(1)} \cos(1.5811t + \phi_1) - 5X_1^{(2)} \cos(2.4495t + \phi_2) \quad (\text{E.7})$$

By using the given initial conditions in Eqs.(E.6) and (E.7), we obtain

$$x_1(t = 0) = 1 = X_1^{(1)} \cos \phi_1 + X_1^{(2)} \cos \phi_2 \quad (\text{E.8})$$

$$x_2(t = 0) = 0 = 2X_1^{(1)} \cos \phi_1 - 5X_1^{(2)} \cos \phi_2 \quad (\text{E.9})$$

$$\dot{x}_1(t = 0) = 0 = -1.5811X_1^{(1)} \sin \phi_1 - 2.4495X_1^{(2)} \sin \phi_2 \quad (\text{E.10})$$

$$\dot{x}_2(t = 0) = -3.1622X_1^{(1)} + 12.2475X_1^{(2)} \sin \phi_2 \quad (\text{E.11})$$

## Example 5.3 Solution

The solution of Eqs.(E.8) and (E.9) yields

$$X_1^{(1)} \cos \phi_1 = \frac{5}{7}; \quad X_1^{(2)} \cos \phi_2 = \frac{2}{7} \quad (\text{E.12})$$

while the solution of Eqs.(E.10) and (E.11) leads to

$$X_1^{(1)} \sin \phi_1 = 0, \quad X_1^{(2)} \sin \phi_2 = 0 \quad (\text{E.13})$$

Equations (E.12) and (E.13) give

$$X_1^{(1)} = \frac{5}{7}, \quad X_1^{(2)} = \frac{2}{7}, \quad \phi_1 = 0, \quad \phi_2 = 0 \quad (\text{E.14})$$

## Example 5.3 Solution

Thus the free vibration responses of  $m_1$  and  $m_2$  are given by

$$x_1(t) = \frac{5}{7} \cos 1.5811t + \frac{2}{7} \cos 2.4495t \quad (\text{E.15})$$

$$x_2(t) = \frac{10}{7} \cos 1.5811t - \frac{10}{7} \cos 2.4495t \quad (\text{E.16})$$

## 5.4 Torsional System

Consider a torsional system as shown in Fig.5.6. The differential equations of rotational motion for the discs can be derived as

$$J_1 \ddot{\theta}_1 = -k_{t1} \theta_1 + k_{t2} (\theta_2 - \theta_1) + M_{t1}$$

$$J_2 \ddot{\theta}_2 = -k_{t2} (\theta_2 - \theta_1) - k_{t3} \theta_2 + M_{t2}$$

which upon rearrangement become

$$J_1 \ddot{\theta}_1 + (k_{t1} + k_{t2}) \theta_1 - k_{t2} \theta_2 = M_{t1}$$

$$J_2 \ddot{\theta}_2 - k_{t2} \theta_1 + (k_{t2} + k_{t3}) \theta_2 = M_{t2} \quad (5.19)$$

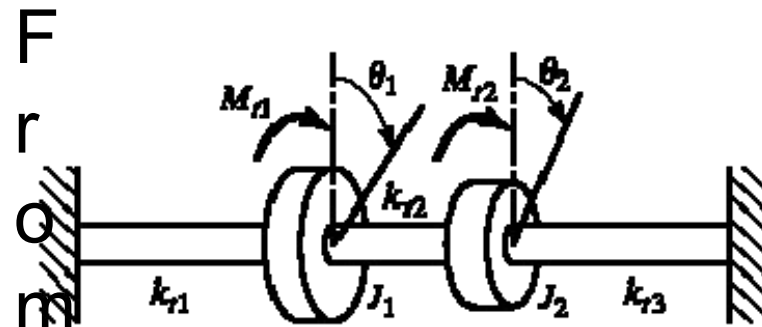
For the free vibration analysis of the system, Eq.(5.19) reduces to



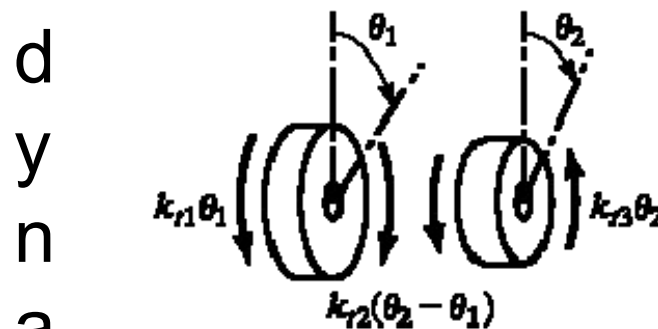
## 5.4 Torsional System

$$J_1 \ddot{\theta}_1 + (k_{t1} + k_{t2})\theta_1 - k_{t2}\theta_2 = 0$$

$$J_2 \ddot{\theta}_2 - k_{t2}\theta_1 + (k_{t2} + k_{t3})\theta_2 = 0 \quad (5.20)$$



(a)



(b)

Figure 5.6: Torsional system with discs mounted on a shaft

## Ex 5.4 Natural Frequencies of a Torsional System

Find the **natural frequencies** and **mode shapes** for the torsional system shown in Fig.5.7 for  $J_1 = J_0$ ,  $J_2 = 2J_0$  and  $k_{t1} = k_{t2} = k_t$ .

*Solution:*

The differential equations of motion, Eq.(5.20), reduce to (with  $k_{t3} = 0$ ,  $k_{t1} = k_{t2} = k_t$ ,  $J_1 = J_0$  and  $J_2 = 2J_0$ ):

$$\begin{aligned} J_0 \ddot{\theta}_1 + 2k_t \theta_1 - k_t \theta_2 &= 0 \\ 2J_0 \ddot{\theta}_2 - k_t \theta_1 + k_t \theta_2 &= 0 \end{aligned} \quad (\text{E.1})$$

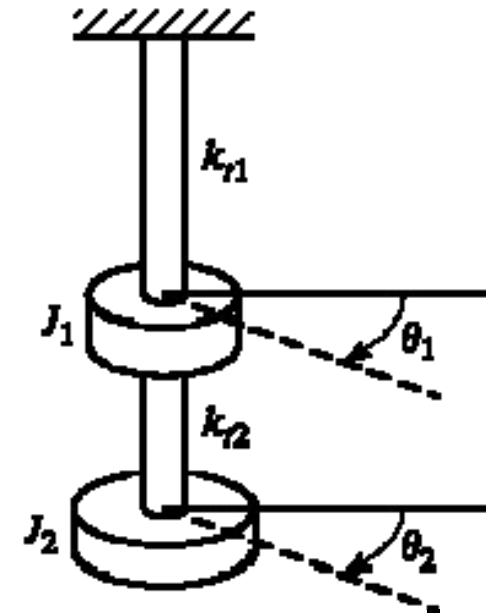


Fig.5.7:  
Torsional system

## Example 5.4 Solution

Rearranging and substituting the harmonic solution:

$$\theta_i(t) = \Theta_i \cos(\omega t + \phi); \quad i = 1, 2 \quad (\text{E.2})$$

gives the frequency equation:

$$2\omega^4 J_0^2 - 5\omega^2 J_0 k_t + k_t^2 = 0 \quad (\text{E.3})$$

The solution of Eq.(E.3) gives the natural frequencies

$$\begin{aligned} \omega_1 &= \sqrt{\frac{k_t}{4J_0} (5 - \sqrt{17})} \quad \text{and} \quad \omega_2 = \sqrt{\frac{k_t}{4J_0} (5 + \sqrt{17})} \quad (\text{E.4}) \\ &= 0.4682\sqrt{k_t / J_0} \quad \quad \quad = 1.5102\sqrt{k_t / J_0} \end{aligned}$$

## Example 5.4 Solution

The amplitude ratios are given by

$$r_1 = \frac{\Theta_2^{(1)}}{\Theta_1^{(1)}} = 2 - \frac{(5 - \sqrt{17})}{4} = 1.7808$$
$$r_2 = \frac{\Theta_2^{(2)}}{\Theta_1^{(2)}} = 2 - \frac{(5 + \sqrt{17})}{4} = -0.2808 \quad (\text{E.5})$$

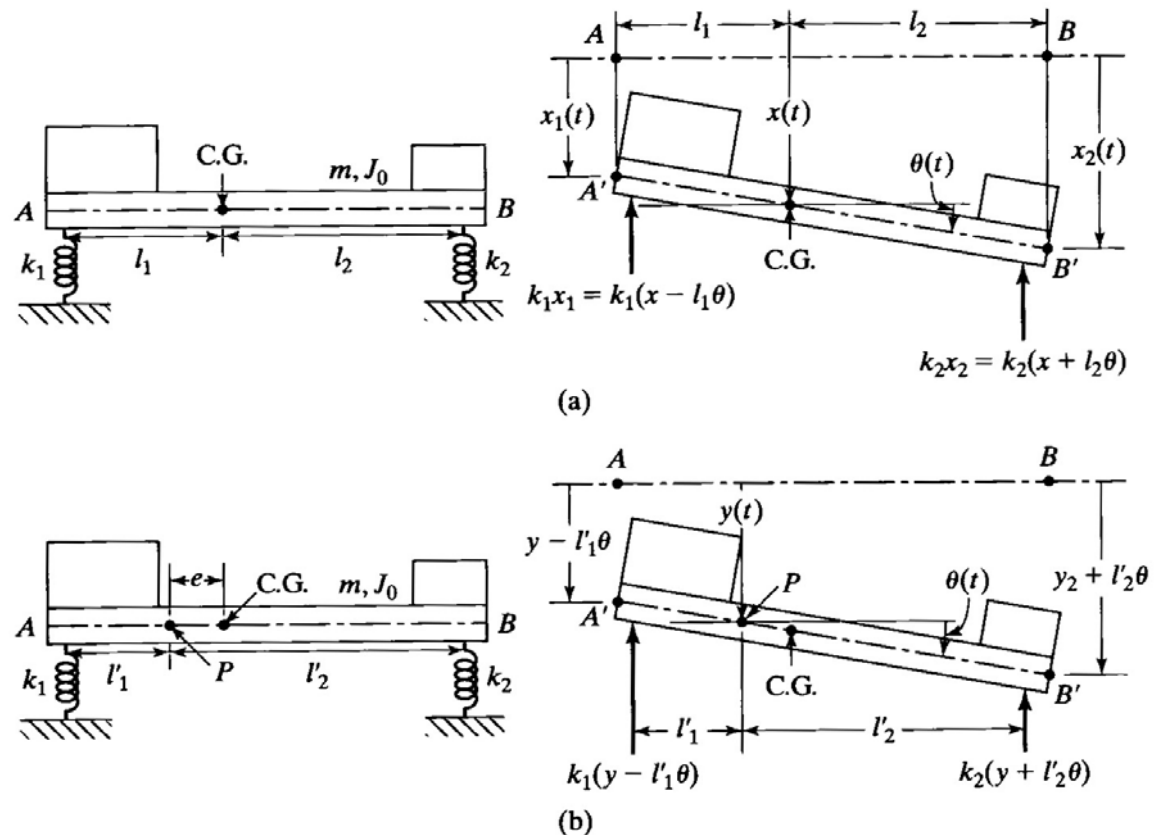
Equations (E.4) and (E.5) can also be obtained by substituting the following in Eqs.(5.10) and (5.11).

$$k_1 = k_{t1} = k_t, \quad k_2 = k_{t2} = k_t,$$
$$m_1 = J_1 = J_0, \quad m_2 = J_2 = 2J_0 \quad \text{and} \quad k_3 = 0$$

## 5.5 Coordinate Coupling and Principal Coordinates

*Generalized coordinates* are sets of  $n$  coordinates used to describe the configuration of the system.

**Equations of motion for a lathe Using  $x(t)$  and  $\theta(t)$**



## 5.5 Coordinate Coupling and Principal Coordinates

From the free-body diagram shown in Fig.5.10a, with the positive values of the motion variables as indicated, the force equilibrium equation in the vertical direction can be written as

$$m\ddot{x} = -k_1(x - l_1\theta) - k_2(x + l_2\theta) \quad (5.21)$$

and the moment equation about C.G. can be expressed as

$$J_0\ddot{\theta} = k_1(x - l_1\theta)l_1 - k_2(x + l_2\theta)l_2 \quad (5.22)$$

Eqs.(5.21) and (5.22) can be rearranged and written in matrix form as

## 5.5 Coordinate Coupling and Principal Coordinates

$$\begin{bmatrix} m & 0 \\ 0 & J_0 \end{bmatrix} \begin{Bmatrix} \ddot{x} \\ \ddot{\theta} \end{Bmatrix} + \begin{bmatrix} (k_1 + k_2) & -(k_1 l_1 - k_2 l_2) \\ -(k_1 l_1 - k_2 l_2) & (k_1 l_1^2 + k_2 l_2^2) \end{bmatrix} \begin{Bmatrix} x \\ \theta \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix} \quad (5.23)$$

The lathe rotates in the vertical plane and has vertical motion as well, unless  $k_1 l_1 = k_2 l_2$ . This is known as *elastic or static coupling*.

### • **Equations of motion Using $y(t)$ and $\theta(t)$ .**

From Fig.5.10b, the equations of motion for translation and rotation can be written as

$$m\ddot{y} = -k_1(y - l'_1\theta) - k_2(y + l'_2\theta) - me\ddot{\theta}$$

## 5.5 Coordinate Coupling and Principal Coordinates

$$J_P \ddot{\theta} = k_1(y - l'_1 \theta)l'_1 - k_2(y + l'_2 \theta)l'_2 - me\ddot{y} \quad (5.24)$$

These equations can be rearranged and written in matrix form as

$$\begin{bmatrix} m & me \\ me & J_P \end{bmatrix} \begin{Bmatrix} \ddot{y} \\ \ddot{\theta} \end{Bmatrix} + \begin{bmatrix} (k_1 + k_2) & (k_2 l'_2 - k_1 l'_1) \\ (-k_1 l'_1 + k_2 l'_2) & (k_1 l'^2_1 + k_2 l'^2_2) \end{bmatrix} \begin{Bmatrix} y \\ \theta \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix} \quad (5.25)$$

If  $k_1 l'_1 = k_2 l'_2$ , the system will have *dynamic or inertia coupling only*.

Note the following characteristics of these systems:



## 5.5 Coordinate Coupling and Principal Coordinates

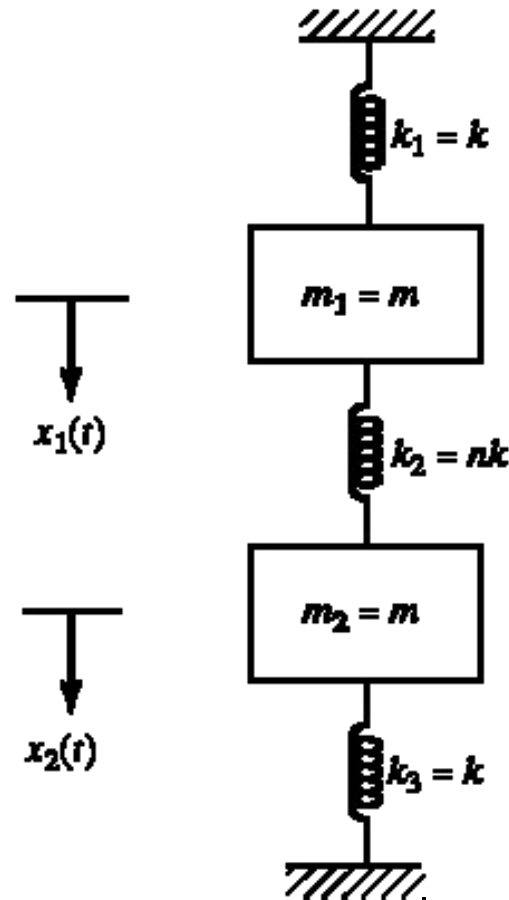
1. In the most general case, a viscously damped 2DOF system has the equations of motions in the form:

$$\begin{bmatrix} m_{11} & m_{12} \\ m_{21} & m_{22} \end{bmatrix} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \end{Bmatrix} + \begin{bmatrix} c_{11} & c_{12} \\ c_{21} & c_{22} \end{bmatrix} \begin{Bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{Bmatrix} + \begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix} \quad (5.26)$$

2. The system vibrates in its own natural way regardless of the coordinates used. The choice of the coordinates is a mere convenience.
3. *Principal or natural coordinates* are defined as system of coordinates which give equations of motion that are **uncoupled both statically and dynamically**.

## Ex 5.6 Principal Coordinates of Spring-Mass System

Determine the principal coordinates for the spring-mass system shown in Fig.5.4.



## Example 5.6 Solution

*Approach:* Define two independent solutions as principal coordinates and express them in terms of the solutions  $x_1(t)$  and  $x_2(t)$ .

The general motion of the system shown is

$$\begin{aligned}x_1(t) &= B_1 \cos\left(\sqrt{\frac{k}{m}}t + \phi_1\right) + B_2 \cos\left(\sqrt{\frac{3k}{m}}t + \phi_2\right) \\x_2(t) &= B_1 \cos\left(\sqrt{\frac{k}{m}}t + \phi_1\right) - B_2 \cos\left(\sqrt{\frac{3k}{m}}t + \phi_2\right)\end{aligned}\quad (\text{E.1})$$

We define a new set of coordinates such that

## Example 5.6 Solution

$$\begin{aligned}q_1(t) &= B_1 \cos\left(\sqrt{\frac{k}{m}}t + \phi_1\right) \\q_2(t) &= B_2 \cos\left(\sqrt{\frac{3k}{m}}t + \phi_2\right)\end{aligned}\tag{E.2}$$

Since the coordinates are harmonic functions, their corresponding equations of motion can be written as

$$\begin{aligned}\ddot{q}_1 + \left(\frac{k}{m}\right)q_1 &= 0 \\ \ddot{q}_2 + \left(\frac{3k}{m}\right)q_2 &= 0\end{aligned}\tag{E.3}$$

## Example 5.6 Solution

From Eqs.(E.1) and (E.2), we can write

$$\begin{aligned}x_1(t) &= q_1(t) + q_2(t) \\x_2(t) &= q_1(t) - q_2(t)\end{aligned}\tag{E.4}$$

The solution of Eqs.(E.4) gives the principal coordinates:

$$\begin{aligned}q_1(t) &= \frac{1}{2}[x_1(t) + x_2(t)] \\q_2(t) &= \frac{1}{2}[x_1(t) - x_2(t)]\end{aligned}\tag{E.5}$$

# Forced Vibration Analysis

The equations of motion of a general 2DOF system under external forces can be written as

$$\begin{bmatrix} m_{11} & m_{12} \\ m_{12} & m_{22} \end{bmatrix} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \end{Bmatrix} + \begin{bmatrix} c_{11} & c_{12} \\ c_{21} & c_{22} \end{bmatrix} \begin{Bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{Bmatrix} + \begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix} \quad (5.27)$$

Consider the external forces to be harmonic:

$$F_j(t) = F_{j0} e^{i\omega t}, \quad j = 1, 2 \quad (5.28)$$

where  $\omega$  is the forcing frequency. We can write the steady-state solutions as

$$x_j(t) = X_j e^{i\omega t}, \quad j = 1, 2 \quad (5.29)$$

# Forced Vibration Analysis

Substitution of Eqs.(5.28) and (5.29) into Eq.(5.27) leads to

$$\begin{bmatrix} (-\omega^2 m_{11} + i\omega c_{11} + k_{11}) & (-\omega^2 m_{12} + i\omega c_{12} + k_{12}) \\ (-\omega^2 m_{12} + i\omega c_{12} + k_{12}) & (-\omega^2 m_{22} + i\omega c_{22} + k_{22}) \end{bmatrix} \begin{Bmatrix} X_1 \\ X_2 \end{Bmatrix} = \begin{Bmatrix} F_{10} \\ F_{20} \end{Bmatrix} \quad (5.30)$$

We defined as in [Section 3.5](#) the mechanical impedance  $Z_{re}(i\omega)$  as

$$Z_{rs}(i\omega) = -\omega^2 m_{rs} + i\omega c_{rs} + k_{rs}, r, s = 1, 2 \quad (5.31)$$

# Forced Vibration Analysis

And write Eq.(5.30) as:

$$[Z(i\omega)]\vec{X} = \vec{F}_0 \quad (5.32)$$

where

$$[Z(i\omega)] = \begin{bmatrix} Z_{11}(i\omega) & Z_{12}(i\omega) \\ Z_{12}(i\omega) & Z_{22}(i\omega) \end{bmatrix} = \text{Impedance matrix}$$

$$\vec{X} = \begin{Bmatrix} X_1 \\ X_2 \end{Bmatrix}$$

$$\vec{F}_0 = \begin{Bmatrix} F_{10} \\ F_{20} \end{Bmatrix}$$

Eq.(5.32) can be solved to obtain:



# Forced Vibration Analysis

$$\vec{X} = [Z(i\omega)]^{-1} \vec{F}_0 \quad (5.33)$$

where the inverse of the impedance matrix is given

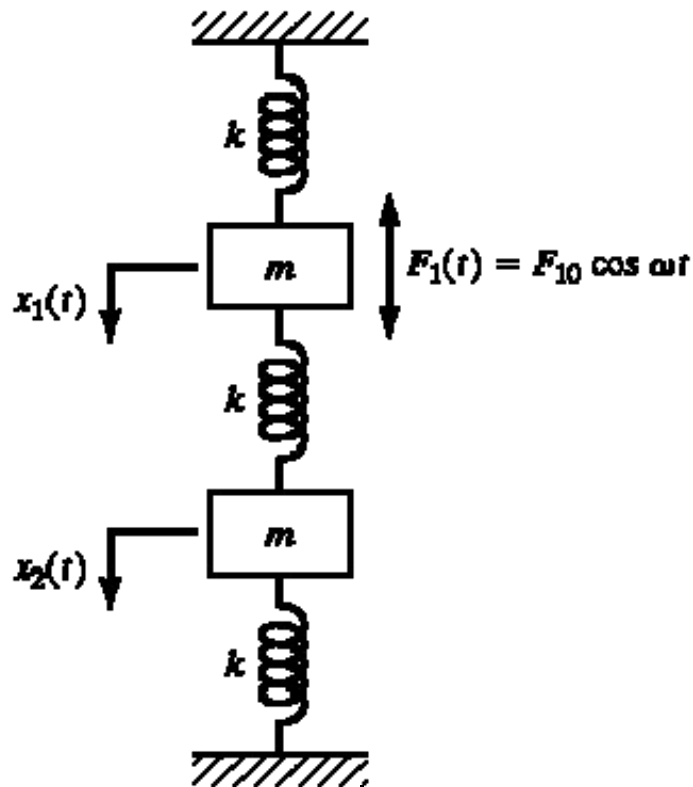
$$[Z(i\omega)]^{-1} = \frac{1}{Z_{11}(i\omega)Z_{22}(i\omega) - Z_{12}^2(i\omega)} \begin{bmatrix} Z_{22}(i\omega) & -Z_{12}(i\omega) \\ -Z_{12}(i\omega) & Z_{11}(i\omega) \end{bmatrix} \quad (5.34)$$

Eqs.(5.33) and (5.34) lead to the solution

$$\begin{aligned} X_1(i\omega) &= \frac{Z_{22}(i\omega)F_{10} - Z_{12}(i\omega)F_{20}}{Z_{11}(i\omega)Z_{22}(i\omega) - Z_{12}^2(i\omega)} \\ X_2(i\omega) &= \frac{-Z_{12}(i\omega)F_{10} + Z_{11}(i\omega)F_{20}}{Z_{11}(i\omega)Z_{22}(i\omega) - Z_{12}^2(i\omega)} \end{aligned} \quad (5.35)$$

## Ex 5.8 Steady-State Response of Spring-Mass System

Find the steady-state response of system shown in Fig.5.13 when the mass  $m_1$  is excited by the force  $F_1(t) = F_{10} \cos \omega t$ . Also, plot its frequency response curve.



## Example 5.8 Solution

The equations of motion of the system can be expressed as

$$\begin{bmatrix} m & 0 \\ 0 & m \end{bmatrix} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \end{Bmatrix} + \begin{bmatrix} 2k & -k \\ -k & 2k \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} F_{10} \cos \omega t \\ 0 \end{Bmatrix} \quad (\text{E.1})$$

We assume the solution to be as follows.

$$x_j(t) = X_j \cos \omega t; \quad j = 1, 2 \quad (\text{E.2})$$

Eq.(5.31) gives

$$Z_{11}(\omega) = Z_{22}(\omega) = -m\omega^2 + 2k, \quad Z_{12}(\omega) = -k \quad (\text{E.3})$$

## Example 5.8 Solution

Hence,

$$X_1(\omega) = \frac{(-\omega^2 m + 2k)F_{10}}{(-\omega^2 m + 2k)^2 - k^2} = \frac{(-\omega^2 m + 2k)F_{10}}{(-m\omega^2 + 3k)(-m\omega^2 + k)} \quad (\text{E.4})$$

$$X_2(\omega) = \frac{kF_{10}}{(-m\omega^2 + 2k)^2 - k^2} = \frac{kF_{10}}{(-m\omega^2 + 3k)(-m\omega^2 + k)} \quad (\text{E.5})$$

Eqs.(E.4) and (E.5) can be expressed as

$$X_1(\omega) = \frac{\left\{ 2 - \left( \frac{\omega}{\omega_1} \right)^2 \right\} F_{10}}{k \left[ \left( \frac{\omega_2}{\omega_1} \right)^2 - \left( \frac{\omega}{\omega_1} \right)^2 \right] \left[ 1 - \left( \frac{\omega}{\omega_1} \right)^2 \right]} \quad (\text{E.6})$$

# Example 5.8 Solution

$$X_2(\omega) = \frac{F_{10}}{k \left[ \left( \frac{\omega_2}{\omega_1} \right)^2 - \left( \frac{\omega}{\omega_1} \right)^2 \right] \left[ 1 - \left( \frac{\omega}{\omega_1} \right)^2 \right]} \quad (\text{E.7})$$

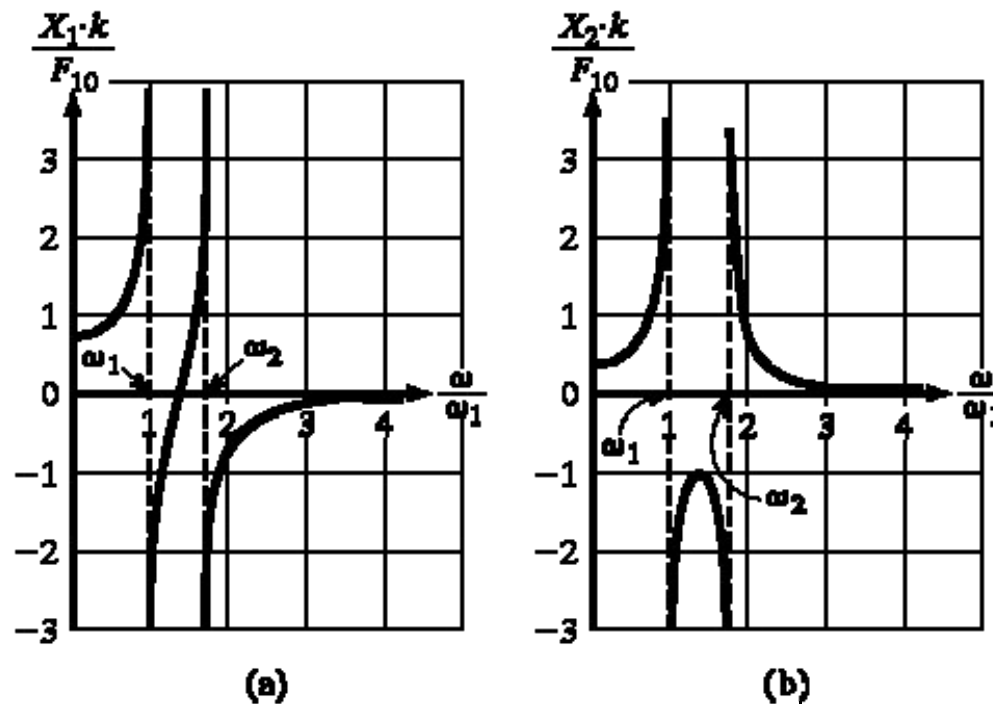


Fig.5.14: Frequency response curves

# Semidefinite Systems

*Semidefinite systems* are also known as *unrestrained* or *degenerate systems*. Two examples of such systems are shown in Fig.5.15. For Fig.5.15a, the equations of motion can be written as

$$\begin{aligned}m_1 \ddot{x}_1 + k(x_1 - x_2) &= 0 \\m_2 \ddot{x}_2 + k(x_2 - x_1) &= 0\end{aligned}\tag{5.36}$$

For free vibration, we assume the motion to be harmonic:

$$x_j(t) = X_j \cos(\omega t + \phi_j), \quad j = 1, 2 \tag{5.37}$$

# Semidefinite Systems

Substituting Eq.(5.37) into Eq.(5.36) gives

$$\begin{aligned}(-m_1\omega^2 - k)X_1 - kX_2 &= 0 \\ -kX_1 + (-m_2\omega^2 + k)X_2 &= 0\end{aligned}\quad (5.38)$$

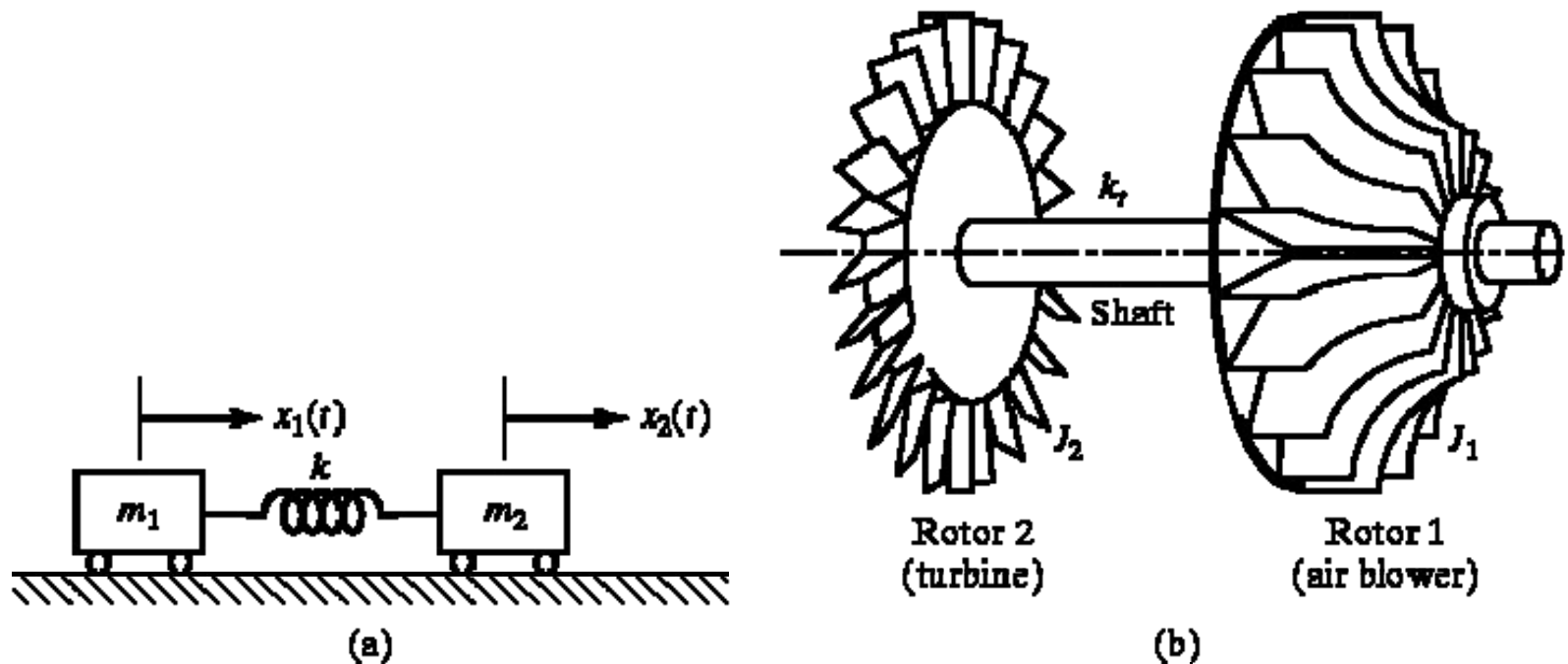


Fig.5.15: Semidefinite Systems

# Semidefinite Systems

We obtain the frequency equation as

$$\omega^2 [m_1 m_2 \omega^2 - k(m_1 + m_2)] = 0 \quad (5.39)$$

From which the natural frequencies can be obtained:

$$\omega_1 = 0 \quad \text{and} \quad \omega_2 = \sqrt{\frac{k(m_1 + m_2)}{m_1 m_2}} \quad (5.40)$$

Such systems, which have one of the natural frequencies equal to zero, are called *semidefinite systems*.



## Frequency (characteristic) equation for 2DOF mass-spring system

$$(m_1 m_2) \omega^4 - \{(k_1 + k_2)m_2 + (k_2 + k_3)m_1\} \omega + \{(k_1 + k_2)(k_2 + k_3) - k_2^2\} = 0 \quad (5.9)$$

$$\omega_1^2, \omega_2^2 = \frac{1}{2} \left\{ \frac{(k_1 + k_2)m_2 + (k_2 + k_3)m_1}{m_1 m_2} \right\} \mp \frac{1}{2} \left[ \left\{ \frac{(k_1 + k_2)m_2 + (k_2 + k_3)m_1}{m_1 m_2} \right\}^2 - 4 \left\{ \frac{(k_1 + k_2)(k_2 + k_3) - k_2^2}{m_1 m_2} \right\}^{1/2} \right] \quad (5.10)$$